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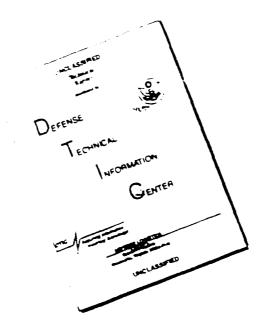
EVALUATION OF MAIL TRAYS

HQ AFMC/LGTP
AIR FORCE PACKAGING EVALUATION ACTIVITY
5215 THURLOW ST
WRIGHT-PATTERSON AFB, OH 45433-5540
JUNE 1993

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PROJECT # 92-119

TITLE: Evaluation of Mail Trays

ABSTRACT

The Air Force Packaging Evaluation Activity (AFPEA), in response to a request from the United States Postal Service Engineering and Development Center (USPS EDC), conducted container testing and materials testing on mail trays and their plastic corrugated material from three manufacturers. The trays were tested against FED-STD-101, Methods 5007.1 and 5019.1 and FED-STD-648A, Paragraph 5.8.3. The material was tested against MIL-P-83668A and MIL-STD-810E, Method 507.3. The mail trays were provided by the USPS EDC. The mail tray material was provided by the manufacturers.

The test series was performed at the Air Force Packaging Evaluation Activity, 5215 Thurlow St, Wright-Patterson AFB, Ohio, 45433-5540.

SUMMARY

An evaluation was done on the performance of the mail trays and their material.

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INTRODUCTION

The United States Postal Service Engineering and Development Center (USPS EDC) requested testing of Standard Type Mail Trays. Testing included container rough-handling tests and materials testing of the mail tray plastic-corrugated material.

The mail trays were supplied by the USPS and the mail tray material was supplied by three manufacturers, identified as Company A, Company B, and Company C, of the mail trays.

The performance of each of the manufacturers' trays and their materials were evaluated. In total, there were four different trays submitted for testing. Company A had two types of containers which will be referred to as "Co A, Container 1" and "Co A, Container 2". Company B had one type of container which will be referred to as "Co B Container". Company C had one type of container which will be referred to as "Co C Container".

CONTAINER AND MATERIAL DESCRIPTION

The United States Postal Service Standard Type Mail Tray is a small, plastic, laminated corrugated container (Figure 1). The container includes a cover, but the significance of the cover is not part of this test report. (Figure 2). Maximum outer container dimensions are 18.30 inches long, 13.31 inches wide, and 11.43 inches deep.

Co A, Container 1 material is a three-layered, corrugated plastic board, where the layers are laminated by glue. Co A, Container 2 material is a five-layered, corrugated plastic board, where the layers are laminated by a fusion bonding process. Co B Container and Co C Container materials are three-layered, corrugated plastic board, where the layers are laminated by heat fusion.

For purposes of this report, Company A material will be denoted as "Co A, Material 1" and "Co A, Material 2". Company B material will be denoted as "Co B Material" and Company C material will be denoted as "Co C Material".

TEST PROCEDURE

The USPS mail trays and their materials were tested in accordance with the Air Force Packaging Evaluation Activity (AFPEA) Test Plan, Project Number 92-119, dated 5 Oct 92. The test plan referenced FED-STD-101C, FED-STD-648B, MIL-P-83668A, and MIL-STD-810E.

The test methods constitute both the procedures for performing the

tests and performance criteria for evaluation of container and container material acceptability. The tests were performed at AFPEA, 5215 Thurlow St, Wright-Patterson AFB, OH 45433-5540.

Test Sequences 2 through 4 of the mail tray container test plan were performed on all four containers with their lids in place. Test Sequence 5 was performed on all four containers without their lids.

Test Sequences 2 and 3 were performed using the same containers. Test Sequences 4 and 5 were performed using the same containers; but, a different set than those used for test sequences 2 and 3.

The containers were inspected for interior and exterior damage after each test sequence. Inspection included container surfaces and structures, and contents (if applicable).

NOTE: The mail tray test plan rough-handling tests are intended to test containers to failure. It is unlikely that most mail trays would receive handling as rough as this in their normal use.

Test Sequences 2 through 7, of the mail tray material test plan, were performed on material sent by the manufacturers; however, because the glued trays were no longer being manufactured, their material samples were cut from existing glued mail tray containers.

The mail tray material was inspected for cracks, delamination, warping, and any visible damage.

Unless otherwise noted, all specimens had been conditioned at $73^{\circ}F$ \pm $5^{\circ}F$ and at 50% relative humidity (RH) \pm 5% RH for a minimum of 24 hours prior to and during testing.

The actual sequence of testing is presented in Appendix A.

CONTAINER IDENTIFICATION

The bottom of each container was labeled with the name of the container's manufacturer. Co A, Container 1 was further identified with "glued" labeled on the bottom and Co A, Container 2 was further labeled with "fusion-bonded" on the bottom. The correlation between labeled and designated container faces is as follows:

Labeled Face 1-2 Left End (Left - Face 3-4) 3-4 Fro t Face (U.S. MAIL) 5-6 Right End (Right of Face 3-4) 7-8 Rear Face (PROPERTY OF U.S.POSTAL SERVICE) BOTTOM TOP Top

The correlation between labeled and designated container corners is as follows:

Labeled Corner	Designated Corner
8T-1T 2T-3T 4T-5T 6T-7T	Top Corner Between Faces 7-8 & 1-2 Top Corner Between Faces 1-2 & 3-4 Top Corner Between Faces 3-4 & 5-6 Top Corner Between Faces 5-6 & 7-8
8B-1B	Bottom Corner Between Faces 7-8 & 1-2
2B-3B	Bottom Corner Between Faces 1-2 & 3-4
4B-5B	Bottom Corner Between Faces 3-4 & 5-6
6B-7B	Bottom Corner Between Faces 5-6 & 7-8

The correlation between labeled and designated container edges is as follows:

Labeled Edge	Designated Edge
1T-2T 3T-4T 5T-6T 7T-8T 1B-2B 3B-4B 5B-6B 7B-8B 1-8 2-3 4-5 6-7	Top Edge Between Corners 1T & 2T Top Edge Between Corners 3T & 4T Top Edge Between Corners 5T & 6T Top Edge Between Corners 7T & 8T Bottom Edge Between Corners 1B & 2B Bottom Edge Between Corners 3B & 4B Bottom Edge Between Corners 5B & 6B Bottom Edge Between Corners 7B & 8B Edge Between Faces 7-8 & 1-2 Edge Between Faces 1-2 & 3-4 Edge Between Faces 3-4 & 5-6 Edge Between Faces 5-6 & 7-8

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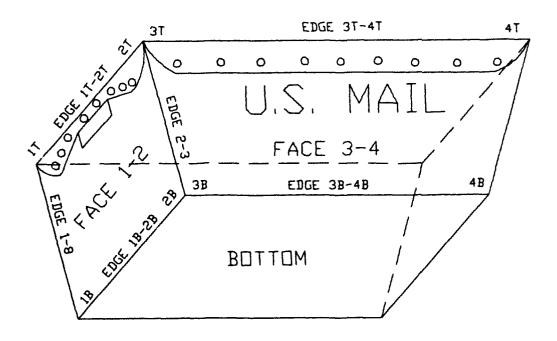
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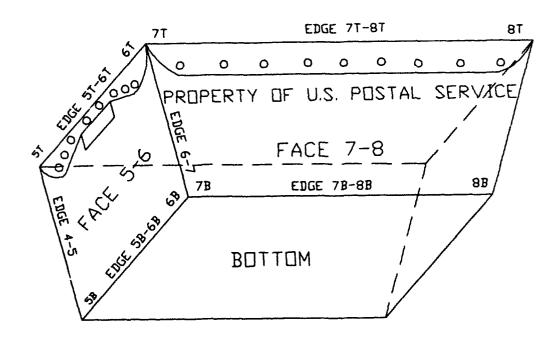
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Sketches 1 and 2 identify the labelled sides, corners and edges as follows:



Sketch 1. Labeled faces, corners and edges on Faces 1-2 and 3-4.



Sketch 2. Labeled faces, corners and edges on Faces 5-6 and 7-8.

TEST SEQUENCES (MAIL TRAY CONTAINERS)

TEST SEQUENCE 1 Measurement and Weight.

A visual inspect: n of the container was made. The container was equipped with one handle on each end of the container. The handles were cut from each end of the container such that a flap of the cut material was folded onto the rest of the material to 1 cm grips.

Container workmanship was visually examined. The container was free of defects that would affect strength, durability, safety or serviceability. Container ultrasonically-fused welds appeared uniform and the container was smooth and free of sharp or jagged edges.

Container color was examined and noted.

The container was not a uniform rectangle in that the top of the container had a larger area than the bottom. Interior and exterior length and width measurements were taken from the top, middle, and bottom of the container. Exterior and interior height measurements were taken.

The following equipment was utilized:

<u>Equipment</u>	<u>Manufacturer</u>	<u>Ser#</u>
Digital Caliper	Mitutoyo	7001683

Three containers from each company were used for taking measurements. A total of five measurements was taken for each dimension. The averages were calculated for each container, and then a grand average was calculated for the length, width, and height for each company.

Abbreviations for the length, width, and height are as follows:

\mathtt{ETL}	Exterior-Top-Length	\mathtt{ITL}	Interior-Top-Length
ETW	Exterior-Top-Width	ITW	Interior-Top-Width
EML	Exterior-Middle-Length	IML	Interior-Middle-Length
EMW	Exterior-Middle-Width	IMW	Interior-Middle-Width
EBL	Exterior-Bottom-Length	IBL	Interior-Bottom-Length
EBW	Exterior-Bottom-Width	IBW	Interior-Bottom-Width
EH	Exterior-Vertical-Height	IH	Interior-Vertical-Height

The average length, width, and height, in inches, for the four containers are as follows:

Manufacturer	ETL	ETW	EML	EMW	EBL	EBW	EH
Co A, Container 1 Co A, Container 2 Co B Container Co C Container	18.18 18.15	13.11 13.11 13.13 13.31	16.52 16.64	11.94 11.90	15.15 15.18	11.11 11.20	11.34 11.43 11.35 11.37
<u>Manufacturer</u>	<u>ITL </u>	ITW	IML	IMW	IBL	IBW	IH

The weight of each manufacturer's container was determined by taking the weight of three containers, from each manufacturer, and averaging them.

The following equipment was utilized:

<u>Equipment</u>	<u>Manufacturer</u>	Mod#	<u>Cal Exp</u>
Beam Balance	OHAUS	1119	22 Nov 93

The weights of the containers are as follows:

Manufacturer		Weight	
Co A, Container	1	805.75	gm
Co A, Container	2	898.33	gm
Co B Container		857.67	gm
Co C Container		829.00	qm

TEST SEQUENCES 2 & 3 - FED-STD-101C, Method 5007.1, 6.3, Procedure A, Free Fall Drop Test.

Equipment	<u>Manufacturer</u>	<u>Model</u>	<u>Ser#</u>	<u>Cal Exp</u>
High Altitude Chamber Drop Tester	Tenney Eng L.A.B.	AD-160	1064018	N/A

The free fall drop tests were conducted in accordance with FED-STD-101C, Method 5007.1. The containers were conditioned at $+140^{\rm OF}$ (test sequence 2) and $-20^{\rm OF}$ (test sequence 3) for 24 hours and then transported to the Conditioning Laboratory where they were loaded with a test load of 30 lb. of white bond paper (Figure 3). Their lids were attached by using strapping tape at both ends of the container, strapping them girthwise. The preparation to fill the containers took approximately 3 minutes once the

container was removed from the chamber. Once the preparation was complete, the containers were dropped in accordance with the above test procedure. To ensure the containers were tested at the correct temperature, only one container was removed from the chamber, prepared with a test load and lid, and dropped at a time.

The containers were dropped 30 inches onto the drop tester steel plate. One drop was made on each face (excluding the top), corner, and edge (Procedure A) for a total of 25 drops. Thirteen drops (four corners, six edges, two sides, and the bottom) were made at +140°F. Twelve drops (using opposite corners, edges, and sides to the 140° F drop) were made at -20° F (Figures 4, 5, and 6).

Visual inspection was noted for each container. The results are as follows:

Free-Fall Drop at +140°F

All of the containers exhibited most of the same characteristics. All containers had rounding of corners and bowing of edges and sides. Each container exhibited slight creasing of the material after various drops (Figures 7, 8, 9, and 10).

Notable differences in performance are listed below.

Co A, Container 1:

1. After drop on corner B4-B5, minor delamination occurred.

Co A, Container 2:

After drop on edge T7-T8, the end of the lid protruded through the handle area on face 5-6 (Figure 11).

Co B Container:

- After drop on edge 1-8, three of the ultrasonically-fused welds separated from face 5-6, disconnecting the right end flap from face 5-6 (Figure 12).
- After drop on face 3-4, the top, right ultrasonically-fused weld on the end flap of face 1-2 separated from face 1-2 (Figure 13).

Co C Container:

- After drop on corner T6-T7, edges 5T-6T and 1T-2T were bent considerably (Figure 14). After drop on edge 4-5, edge 5T-6T was bent further.
- After drop on edge T7-T8, the end of the lid popped through the handle area on face 5-6 (Figure 15).

Note: When Co A, Container 1 was removed from the environmental chamber, it was apparent that it had increased in yellow coloration due to accelerated aging of the glue.

Co A, Container 1, Co A, Container 2, and Co C Container appeared usable after the high-temperature drops; however, Co B Container was beginning to disassemble and may have become inadequate for its normal use.

The damage to Co B Container which resulted during these drops is classified as a major defect according to USPS-T-1155C, <u>United States Postal Service Specification Test Procedures for Polyethylene Mail Tray</u>.

Free-Fall Drop at -20°F

The same containers were used for the cold-temperature drops, utilizing the corners, edges, and faces not used for the high-temperature drops.

All of the containers exhibited most of the same characteristics. All containers had rounding of corners and bowing of edges and sides. Each container exhibited slight creasing of the material after various drops (Figures 16, 17, 18, 19).

Notable differences in performance are listed below.

Co A, Container 1:

- After drop on corner T4-T5, the end of the lid protruded through the handle area on face 5-6 (Figure 20).
- After all drops were completed, the lower, right ultrasonically-fused weld on the end flap of face 1-2 had separated from face 1-2 (Figure 21).

Co A, Container 2:

 After drop on edge T3-T4, the end of the lid protruded through the handle area on face 5-6 (Figure 22).

Co B Container:

- 1. After drop on corner B6-B7, the right end flap on face 5-6 was separated further from face 5-6 (Figure 23).
- 2. After drop on corner T4-T5, the end of the lid protruded through the handle area on face 5-6 (Figure 24).
- After drop on edge T1-T2, the top, left ultrasonically-fused weld on the end flap of face 1-2 became loose.
- 4. After drop on side 5-6, the left, middle and left, bottom ultrasonically-fused welds on the end flap of face 1-2 separated from face 1-2 (Figure 25).

Co C Container:

- 1. After drop on corner T1-T8, the top, right ultrasonically-fused weld on the end flap of face 1-2 became loose.
- 2. After drop on corner T4-T5, the plastic cracked in the middle of edge 5T-6T, and the top left and right ultrasonically-fused welds on the end flap of face 5-6 became loose (Figure 26).
- 3. After drop on edge T3-T4, the steel rod in edge 5T-6T broke in half at a weld joint and protruded through the plastic (Figure 27).

Co A, Container 1 and Co A, Container 2 appeared usable after the low-temperature drops; however, Co B Container had disassembled further and may have become inadequate for its normal use. Also, Co C Container became unusable due to the protruding metal from the top of the handle on face 5-6.

According to USPS-T-1155C, Co A, Container 1, Co B Container, and Co C Container developed major defects during these drops.

After both the high-temperature and low-temperature drops were completed, Co A, Container 1 and Co A, Container 2 appeared usable. Co B Container's use appeared questionable and Co C Container was unusable.

TEST SEQUENCE 4 - FED-STD-101C, Method 5019.1, <u>Vibration</u> (Repetitive Shock) Test.

<u>Equipment</u>	Manufacturer	<u>Model</u>	<u>Ser#</u>	<u>Cal Exp</u>
Vibration Machine	L.A.B. CORP.	5000-96 в	56801	N/A

The test was conducted in accordance with FED-STD-101C, Method 5019.1, at ambient temperature. Containers used for the drop tests were replaced; new containers were used for the Repetitive Shock Test.

The four containers were loaded with 30 lb. of sand and lids attached in the same manner as in the drop tests (Figure 28). The containers were placed on the vibration table. Restraints were utilized that would prevent the containers from sliding off the table or sliding into each other (Figures 29 and 30). The containers were allowed unrestricted movement from the centered position on the table about 1/2 inch in any horizontal direction.

The table frequency was increased from 0.0 Hertz (Hz) until the container left the table surface. At 4.6 Hz input vibration frequency, one inch double amplitude, a 1/16 inch thick bar could be slid freely between table and container under all points of the container. This condition was maintained for a period of two hours.

Visual inspection revealed no damage to any of the containers or their contents (Figures 31, 32, 33, and 34).

TEST SEQUENCE 5 - MIL-STD-648B, 5.8.3 <u>Hoisting Fittings Strength</u> Test.

The following equipment was utilized:

<u>Equipment</u>	<u>Manufacturer</u>	<u>Model</u>	<u>Ser#</u>	<u>Cal Exp</u>
Hoist	Coffing	3 Ton	SRD-112-CP	N/A

Containers used for the Repetitive Shock test were also used for the Hoist Fitting Strength Test. A 156.25 pound load, representing at least five times the 31.25 pound gross container weight, was placed on each container. The test load consisted of 156.25 pounds of bags of sand.

The containers were lifted, one at a time, completely off the ground for 5 minutes utilizing each handle (Figure 35).

All of the containers exhibited most of the same characteristics. Due to the pressure of the hoist strap and the weight of the load, all containers had their handles bowed inward, their sides bowed outward, and had some wrinkling of the material (Figures 36, 37, 38, and 39).

Notable differences in performance are listed below.

Co B Container:

1. The inner, right ultrasonically-fused weld, when looking at the handle from the inside of the container, on the handle of face 5-6, separated from face 5-6 (Figure 40).

All containers appeared usable after the test.

TEST SEQUENCES (MAIL TRAY MATERIAL)

TEST SEQUENCE 1 - Types of Material.

See paragraph, Container and Material Description, page 1.

TEST SEQUENCE 2 - MIL-P-83668A, paragraph 4.5.4, Basis Weight.

Equipment	Manufacturer	<u>Ser#</u>	<u>Cal Exp</u>
Digital Caliper	Mitutoyo	7001683	N/A
Beam Balance	OHAUS	1119	22 Nov 93

Ten test specimens were cut to 10 in. x 10 in., for three of the four materials, from 25 in. x 25 in. plastic board sheets. Due to the inability to obtain 25 in. x 25 in. plastic board sheets of Co A, Material 1 material, ten 10 in. x 10 in. specimens were cut from the bottom of Co A, Container 1 containers. Basis weight, the ratio of mass per area, was then calculated for each specimen. The Basis Weight was calculated as the average of the ten specimens.

Basis Weight measurements were calculated for use in the Flute T-Peel Strength Test only. Individual Basis Weight measurements were not compared to the Basis Weight specified by the manufacturers.

<u>Material</u>	<u>Basis Weight</u>
Co A, Material 1	0.152 lb./ft ²
Co A, Material 2	0.160 lb./ft^2
Co B Material	0.171 lb./ft^2
Co C Material	0.162 lb./ft^2

TEST SEQUENCE 3 - MIL-P-83668A, paragraph 4.5.5, Thickness.

Equipment	<u>Manufacturer</u>	Ser#
Digital Caliper	General/MG	272350

An overall thickness of each material was calculated, by the following method, solely for comparing the materials to each other. Thickness measurements were not compared to manufacturer-specified thicknesses.

Ten specimens, 3 in. x 5 in., for three of the four materials, were cut from 25 in. x 25 in. plastic board sheets. Due to the inability to obtain 25 in. x 25 in. plastic board sheets of Co A, Material 1 material, ten 3 in. x 5 in. specimens were cut from the sides of Co A, Container 1 containers. The thickness of each specimen was measured at five non-overlapping points. The thickness of each specimen was calculated as the average of the five measurements. The thickness of each material was calculated as the average of the ten specimen averages.

Material	Thickness
Co A, Material 1	0.143 in.
Co A, Material 2	0.162 in.
Co B Material	0.157 in.
Co C Material	0.171 in.

TEST SEQUENCE 4 - MIL-P-83668A, paragraph 4.5.6, <u>Bending</u> Oualities.

Two test specimens, 10 in. \times 10 in., from each material, were scored parallel and perpendicular to the direction of the corrugated medium (Figure 41). The test specimens were then folded 180° toward one facing and 180° toward the other along the scoreline both parallel and perpendicular to the direction of the corrugated medium. The specimens were inspected for breaks or cracks after folding. Cracks or breaks occurring at the intersection of score lines were discarded.

No visible damage occurred to any of the specimens.

TEST SEQUENCE 5 - MIL-P-83668A, paragraph 4.5.7, <u>Flute T-Peel</u> <u>Strength Test</u>.

Equipment	<u>Manufacturer</u>	Ser#	<u>Cal Exp</u>
Tensile-Compression Tester	Instron	1597	N/A
Tensile Load Cell	Instron	A414	N/A
Data Acquisition System	N/A	N/A	N/A

Ten test specimens, 11 in. long, were cut from 25 in. x 25 in. sheets of each material. Due to the lack of glued material, the glued samples were cut from the bottom of existing mail trays. The width of each specimen was cut such that each facing included exactly five flute peaks. A band saw was used to separate the facings at one end to a length of 3 in. by cutting through the corrugations, being careful not to damage the facings. Using a tension testing machine, a load was applied at a constant rate of 1 in. per minute along a line parallel to the specimen length (figures 42 and 43). The test was continued for a minimum of 2.5 in. of separation. Due to constraints of the tension testing machine, the length of separation, stated in MIL-P-83668A, was decreased from 6 in. to 2.5 in. The change did not appear to have any significance on the test results.

The load recorded for peel strength is the load recorded after the initial peak is reached (i.e., when the load recorded reflects the force required for separation) or the load required for the one of the facings to rupture.

The basis weight of each material required a peel strength of 35 pounds.

Co A, Materials 1 and 2, and Co C Material had peel strengths of under 35 lb:

- Co A Material 1: Peel Strength measured 5 lb.
- Co A Material 2: Peel Strength measured 2 lb.
- Co C Material : Peel Strength measured 10 lb.

Co B Material did not peel. It appeared as though the bonding of the layers was so strong that one facing would rupture before peeling would start (Figure 44). This occurred at 58 lb.

The initial peak force measured for Co A, Material 1 was 21 lb. The initial peak force measured for Co A, Material 2 was 39 lb. The initial peak force measured for Co C Material was 32 lb.

TEST SEQUENCE 6 - MIL-STD-810E, method 507.3, Humidity Test.

Equipment	<u>Manufacturer</u>	<u>Ser#</u>	<u>Cal Exp</u>
Temperature and Humidity Test Chamber	Blue-M	FRP103	30 Sep 93

Ten specimens from each material were cut to 5 in. x 5 in. and conditioned at $+140^{\circ}F \pm 5^{\circ}F$ and 90% RH \pm 5% RH for 168 hours. The samples were then inspected for any visible damage such as warping or delamination.

The results are as follows:

Co A, Material 1

When the samples were removed from the humidity chamber, it was apparent that their color had yellowed considerably. Six of the ten samples had started to delaminate (Figure 45). Of the four samples which had not started to delaminate, two were easily delaminated by pulling the layers apart by hand, using little effort. (Figure 46). The two remaining samples were not physically separated in order to determine if the glue would hold up after completely cooling and drying. The two remaining samples could not be delaminated by hand after cooling and drying.

All of the samples had warped slightly.

Co A, Material 2

All of the samples had warped slightly, but no other visible damage had occurred (Figure 47).

Co B Material

All of the samples had warped considerably, but no other visible

damage had occurred (Figure 48).

Co C Material

All of the samples warped slightly. Of all of the materials, this material warped the least. No other visible damage had occurred (Figure 49).

NOTE: The Flatness Test, MIL-P-83668A, paragraph 4.5.8, was not performed due to the inability to obtain new Co A, Material 1 material.

CONCLUSION

The mail tray tests and mail tray material tests were conducted solely for comparison purposes. The tests were not used to either pass or fail the mail trays or mail tray material. To assist with this comparison, the following events are highlighted:

The container drop tests resulted in three of the four mail trays acquiring major defects according to USPS-T-1155C. The only mail tray which did not acquire a major defect was Co A, Container 2.

The container vibration test resulted in no damage to any of the containers.

The Hoist-Fittings Strength Test resulted in one tray, Co B Container, acquiring a major defect according to USPS-T-1155C.

The Basis Weight was determined for comparison purposes and for use in the Flute T-Peel Strength Test. The Basis Weight was not compared to the manufacturer-specified basis weight.

The Thickness test was used for comparison purposes only.

The Bending Qualities test resulted in no damage to any of the material specimens.

The Flute T-Peel Strength Test resulted in only one specimen, Co B Material, meeting the specified requirement of a 35 lb. peel strength.

The Humidity Test resulted in slight warping of all of the specimens. In addition, Co A, Material 2 delaminated considerably.

A summary of the mail tray test results and the mail tray material test results are located on the following pages, in Table 1 and Table 2, respectively.

TABLE 1 - Tray Summary

					HOIST FITTINGS
CONTAINER	WEIGHT	DROP TEST, +140°F	DROP YEST, -25°F	VIBRATION TEST	STRENGTH TEST
	1				
Co A, Container 1	805.75 gm 1.	- 1	1. Rounded corners	No damage	1. Handles bent inward
			2. Bowed edges and sides		2. Sides bowed outward
		3. Slight material creasing	3. Slight material creasing		3. Wrinkling of material
			4. Lid protrusion through		
			handle area		
			5. Total of one separated		
			weld		
Co A, Container 2	898.33 gm	898.33 gm 1. Rounded corners	1. Rounded corners	No damage	1. Handles bent inward
		2. Bowed edges and sides	2. Bowed edges and sides		2. Sides bowed outward
		3. Slight material creasing	3. Slight material creasing		3. Wrinkling of material
		4. Lid protrusion through	4. Lid protrusion through		
		handle area	handle area		
Co B Container	857.67 gm 1.	- 1	1. Rounded corners	No damage	1. Handles bent inward
					2. Sides bowed outward
		3. Slight material creasing	3. Slight material creasing		3. Wrinkling of material
		4. Total of four separated	4. Lid protrusion through		4. Total of one separated
		welds	handle area		weld
			5. Total of two separated		
			welds		
Co C Container	820 OD Gm	829 Of am 1 Pounded comore		7 (14	-
		2. Bowed edges and sides	2 Bowled edges and sides	en dellande	1. Handles bent inward
		3. Slight material creasing	3. Slight material creasing		3. Wrinkling of material
•		4. Two edges bent	1		
	•		joint		
		5. Lid protrusion through			
		handle area			

TABLE 2 - Material Summary

MATERIAL					
	BASIS WEIGHT	THICKNESS	BENDING QUALITIES	STRENGTH TEST	HUMIDITY TEST
Co A. Material 1	0.152 lb./sq-ft	0.1429 in.	No damage	5 lb.	1. Slight warping
					2. Delamination
Co A, Material 2	0.160 lb./sq-ft	0.1621 in.	No damage	2 lb.	Slight warping
Co B Material	0.171 lb./sq-ft	0.1569 in.	No damage	10 lb.	Slight warping
	4 1, 1, 07, 0	21112		ă T	Slight Warning

RECOMMENDATIONS

- 1. The glued tray material delaminated substantially during the Humidity test. Recommend that the glued mail trays not be utilized in high-temperature, high-humidity environments. Recommend that USPS-T-1154B, <u>United States Postal Service Specification Test Procedures for Polyethylene Mail Tray Material</u>, paragraph 3.1, be worded such that the trays be manufactured by either a heat-fusion process or the fusion-bonding process.
- 2. The Flute T-Peel Strength requirement of 35 lb. appeared to have little effect on mail tray performance. Recommend the test methods outlined in the mail tray test plan on page 21 be used as first article testing for future mail tray procurements.

APPENDIX A
TEST SEQUENCES

_	AIR FORCE		ontainer T			3						
COL	NTAINER SIZE	1 v W v f)) (INCHES)	WEIG	HT (L	RS)	CUR	F (CII	FT \	92-11 QUANTITY		
	INTERIOR:		TERIOR:	GROSS:		ITEM:	000	£ (00.	· · · · · ·	COART	52.5	
As	noted	As no	ted	As	not	ed	As	note	ed	4	05 OC	Г 92
EM NA	ME						UFACT					
						As	note	ed				
ONTAL	NER NAME					<u> </u>			CO	NTAINER C	OST	
Mai	l Tray								ţ			
	ESCRIPTION				_					·		
Pla	stic board	i (Type	: II - Lan	minated	d Co	rrug	ated)				
ONDIT	IONING		· · · · ·			-						-
Amb	ient testi	ing unl	ess other	rwise s	spec	ifed						
rest	REF STD/S	PEC			<u> </u>			ſ	CON	TAINER	INSTR	 •
NO.	AND TEST ME		TEST TI	TLE AND	PARA	AETER	S			TATION	MENTAT	_
												
1.	MEASUREME	ENT AND	WEIGHT]				
A.	Co A, Cor	ntainer	1 - glu								İ	
1	. Exterior	3	Top:	18.16								
			Middle:									
			Bottom: Height:			ХІ	1.13	111.	(τ	x w)		
2	. Interior	:	Top: Middle:	16.98								
								E .	•	•		
l			Bottom: Height:			Х	0.95	ın.	(T	x w)	}	
			neight.	11.21	111.							
3	. Maximum	Cube:	1.57 ft	5								
	tto i arb b	Th am a	1 70	11-								
4	. Weight	Item: Gross										
		01050] 31.70									
	Co A, Cor											
긔	. Exterior	-	Top:	18.18								
-			Middle: Bottom:									
			Height:			V 1	1.11	1111	(1	л w)		
2	. Interior	:	Top:	17.02								
			Middle:								1	
			Bottom: Height:			ХІ	0.80	ın.	(т	x w)		
			To Lync.	11.50	T11 •							
3	. Maximum	Cube:	1.58 ft	3								
أم	Wainht	T+c	1 00	114							1	
4	. Weight	Item: Gross	I .									
		31035	31.96	T.5/							İ	
								İ				
OMME	NTC.							<u> </u>				
, UMME	M I 3:											
REPA	RED BY:									. Wood ering,		

AIR FORC				TIO	N A	CTIV	ITY	ļ	AFPEA	PROJE	CT NUMB	EK
	(C	ontainer To	est Plan)					ļ		-119		
CONTAINER SIZE INTERIOR:		(INCHES) TERIOR:	WEIGH GROSS:		BS) TEM:	CUBE	(CU. F	T.)	QUANT	rity	DATE	
As noted	As n	oted	As	not			note	d		4	05	OCT 9
M NAME				Ì	MANU	JFACTU						
						As n	oced					
ITAINER NAME							-	CO	NTAINE	R COST	•	
Mail Tray							1					
Plastic boa	rd (Typ	e II - La	minate	i Co	rruc	ated)					
DITIONING		- 				<u></u>						
Ambient tes	ting un	less othe	rwise «	enec	ifia	he						
REF STD/		ress och	TWISE .	spec	1110				AINER		INST	
AND TEST MI PROCEDUR		TEST TI	TLE AND P	ARAM	ETERS	5	•		TATION		MENT	
		D WEIGHT				[
C. Co B Co 1. Exteri			iused, 3 18.15			12 12	:	/1	\			
1. Exceri	OF	Top: Middle:	16.64	in.	х . х 1	11.90	in.	(1	x w)	Ì		
		Bottom:	15.18	in.	x 1	11.20	in.	(ī	x w)	Ì		
		Height:	11.35	in.]						
2. Interi	or	Top:	16.96	in.	x 1	12.19	in.	(1	x w)			
		Middle:	15.85	in.	\mathbf{x}	11.81	in.	(1	x w)	ļ		
		Bottom:			x 1	10.91	in.	(1	x w)			
		Height:	11.21	ın.								
3. Maximu	m Cube:	1.57 ft	3									
4. Weight	Item	1.89) lb							ļ		
	Gros	s: 31.89) lb									
D. Co C Co	ntainer	- heat i	fused.	3 -1 a	ver	į				{		
1. Exteri		Top:	18.30			13.31	in.	(1	x w)	ļ		
		Middle:										
		Bottom: Height:				10.93	ın.	(т	x w)	Ì		
		nergne.	11.57									
2. Interi	or	Top:	17.12									
		Middle: Bottom:										
}		Height:				20.30	111.	(-	~ " <i>,</i>			
3. Maximu	m Cube:	_	_									
4. Weight	Item	1.83	1 1 1 1 1									
4. Weight	Gros											
MMENTS:	<u> </u>						-					
EPARED BY:							Larr					
Susan J. Mi	sra, Ma	terials I	Enginee	r	Mate	erial	s Eng	ine	erir	ng, AI	FPEA	

	AIR FORCE				ION A	CTIV	ITY	AFPEA PR	OJECT NUMBE	R
		(C	ontainer T	est Plan)				92-119	9	
	NTAINER SIZE ((LBS)	CUB	E (CU. FT.	QUANTITY	DATE	
	INTERIOR:		TERIOR:	GROSS:	ITEM:					
AS :	noted	As no	ted	As n	oted		noted	4	05 OCT	<u> 92</u>
EM N	AME					FACTU note				
ONTA	INER NAME					· · ·	C	ONTAINER C	OST	
	l Tray	<u></u>								
	DESCRIPTION stic board	(Type	II - Lar	ninated (orruga	+64\				
	IONING	(1)pe		armaceu (
	ient testi	ng unle	ess other	cwise spe	ecified					
EST	REF STD/SI			wibe bpt		·				
NO.	AND TEST MET PROCEDURE		TEST TI	TLE AND PA	RAMETERS			NTAINER INTATION	INSTR MENTAT	_
2.	ROUGH HAN	חודאכי	מיים הביים (דו	140°E)						
۷٠	FED-STD-1			ll drop t	test		One dro	n on	Visual	
	Method 50			ight: 30			each f		inspect	ion
	Rectangula			on at +14			face (inopect.	- 0
	Level B			inimum of			top),			
	Procedure	A	hours.	Load wit	h 30 l		and cor			
			test.				Total odrops.	of 25		
3.	ROUGH HAN	DLING '	TESTS (-20°F) Free fall drop test.							
	FED-STD-1				test.	Ì	One dro	no ac	Visual	
	Method 50	07.1		ight: 30			each fl		inspect	ion
	Rectangula	ar		on at -20		F	face (except	_	
	Level B	_		inimum of			top), e			
	Procedure	A		Load wit	th 30 1		and con			
			test.				Total d	of 25		
4.	REPETITIV	E SHOC	K TEST				drops.			
	FED-STD-1			citation	of				Visual	
	Method 50	19.1		double an		e.			inspect	icn
				cy deter						
				in. max:						
				ce from the contract of the co						
			Two hour		test.	;				
5.	HOISTING	FITTING	 \$S STRENG	TH TEST	•					
	FED-STD-6			6.25 lb					Visual	
	Para. 5.8	. 3	in the 1	loaded co	ontaine				inspect	ion
				f the flo						
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00000	MTC									
OMME	ints: T NOs. 5 ai	nd 6	ROUGH HAN	אחו.דאב ייי	7 Subs	1 4 001	F) and	(-20°E) •	Note	
tha	t the drop	heiah	t has bee	en change	ed due	to s	tandaro	(-20'F): lusage (Note of contain	ner
REPA	RED BY:							A. Wood		
Sus	an J. Misra	a, Mate	erials En	ngineer				ering, A	AFPEA	

AIR FORCE PACKAGING	CTIVITY	AFPEA PROJECT NUMBE			
(Container	Test Plan)			92-1	19
CONTAINER SIZE (L x W x D) (INCHES) INTERIOR: EXTERIOR:	WEIGHT GROSS:	(LBS)	CUBE (CU. F	T.) QUANTITY	DATE
As noted As noted			1	4	05 OCT 9
EM NAME Mail Tray Material		1	FACTURER		
ONTAINER NAME		AS	noted		
UNIAINER NAME				CONTAINER CO)ST
ACK DESCRIPTION Plastic board (Type II -	Laminated	Corrug	gated)		
ONDITIONING					
Ambient testing unless of	herwise s	pecifie	ed		
REF STD/SPEC AND TEST METHOD OR PROCEDURE NO'S	TITLE AND PAI	RAMETERS		ONTAINER IENTATION	INSTRU- MENTATION
1. MANUFACTURERS - TYPES	OF MATER	IAL			
A. Co A, Material 1 - gl	ued, 3 la	yer			
B. Co A, Material 2 - fu		•	ayer		
c. Co B Material - heat		-			
D. Co C Material - heat	fused, 3-	layer			
Paragraph 4.5.4 Minim		e test	a.		Measuremen
plast	from the	sheet w			
10 ir	um dimens	•			
l l	tion at +'	_	(
	± 5% RH f		ulty		1
minim	um of 24 1	nours.			
			Ì		
}					
OMMENTS:					1
-					
REPARED BY:		APPROVE	D BY: Larry	, A Mand	
Susan J. Misra, Materials	Engineer		rials Engi		AFPEA

CONTAINER SIZE (L x W x D) (INCHES) WEIGHT (LBS) CUBE (CU. FT.) QUANTITY DATE INTERIOR: EXTERIOR: ITEM:	AIR FORCE PACKAGING EVALUATION ACTIV								AFPEA PROJECT NUMBER			
INTERIOR: EXTERIOR: GROSS: ITEM: As noted As noted As noted As noted As noted MANUFACTURER As noted CONTAINER NAME Plastic board (Type II ~ Laminated Corrugated) CONDITIONING Ambient testing unless otherwise specified TEST NO. THEST WILL-P-83668A Paragraph 4.5.5 minimum dimensions of 3 in. x 5 in., shall be cut from each sheet of plastic board. The thickness of each test specimen shall be measured at 5 non-overlapping points. Condition at +70°F ± 5°F and 50°k RH ± 5% RH for a minimum of 24 hours. 4 BENDING QUALITIES MIL-P-83668A Paragraph 4.5.6 11 in. x 11 in. shall be scored parallel and perpendicular to the direction of the corrugated medium. The test specimen shall then be folded 180° toward one facing and 180° toward the other along the scoreline both parallel and perpendicular to the direction of the corrugated medium. Condition at +70°F ± 5°F and 50°k RH ± 5% RH for the corrugated medium. The test specimen shall then be folded 180° toward one facing and 180° toward the other along the scoreline both parallel and perpendicular to the direction of the corrugated medium. Condition at +70°F ± 5°F and 50°k RH ± 5% RH for												_
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Mail Tray Material As noted CONTAINER NAME ARCK DESCRIPTION Plastic board (Type II - Laminated Corrugated) CONDITIONING Ambient testing unless otherwise specified TEST MEF STD/SPEC MO. AND TEST METHOD OR PROCEDURE NO'S THICKNESS MIL-P-83668A Paragraph 4.5.5 Paragraph 4.5.5 MIL-P-83668A Paragraph 4.5.5 MIL-P-83668A Paragraph 4.5.5 MIL-P-83668A Paragraph 4.5.5 MIL-P-83668A Paragraph 4.5.5 MIL-P-83668A Paragraph 4.5.6 MENDING OUALITIES MIL-P-83668A Paragraph 4.5.6 MIL-P-83668A Paragraph 4.5.7 MIL-P-83668A Paragraph 4.5.5 M			As n	oted			<u> </u>		4	05	OCT	9
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and 50% RH <u>+</u> 5% RH for										į		
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due to physical limitations.	due	e to physi										
Susan J. Misra, Materials Engineer Materials Engineering, AFPEA												_

AFALD FORM 4

AFPEA PROJECT NUMBER AIR FORCE PACKAGING EVALUATION ACTIVITY (Container Test Plan) 92-119 CONTAINER SIZE (L x W x D) (INCHES) WEIGHT (LBS) CUBE (CU. FT.) QUANTITY DATE GROSS: ITEM: INTERIOR: **EXTERIOR:** As noted As noted 05 OCT 92 ITEM NAME MANUFACTURER Mail Tray Material As noted CONTAINER NAME CONTAINER COST PACK DESCRIPTION Plastic board (Type II - Laminated Corrugated) Ambient testing unless otherwise specified REF STD/SPEC TEST INSTRU-CONTAINER AND TEST METHOD OR PROCEDURE NO'S TEST TITLE AND PARAMETERS NO. **ORIENTATION** MENTATION 5. FLUTE T-PEEL STRENGTH MIL-P-83668A Five test specimens shall Visual and Paragraph 4.5.7 be cut from each of two dimensional plastic board sheets so inspection that no portion of the test specimens is within 3 in. of any of the sheet edges. Cut the width to include exactly five flute peaks on each facing. The test specimens shall be 11 in. long. Condition at $+70^{\circ}F \pm 5^{\circ}F$ and 50% RH + 5% RH for a minimum of 24 hours. Using a tension testing machine, apply a load at a constant rate of 1 in. per minute along a line parallel to the specimen length.

COMMENTS:

TEST NO. 5, FLUTE T-PEEL STRENGTH: Note that specimen length has been

The test shall be

continued for a minimum of 6 in. of seperation or until one of the facings ruptures.

changed due to physical limitations.

PREPARED BY:

Susan J. Misra, Materials Engineer

APPROVED BY: Larry A. Wood
Materials Engineering, AFPEA

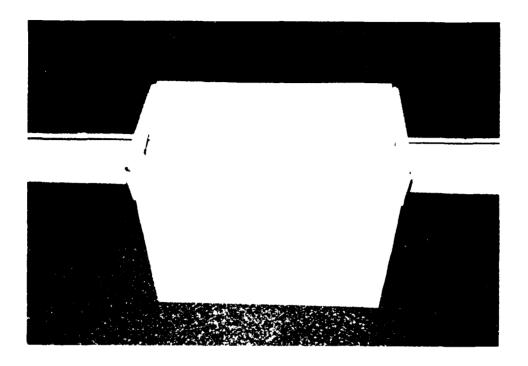
AFPEA PROJECT NUMBER AIR FORCE PACKAGING EVALUATION ACTIVITY (Container Test Plan) 92-119 CUBE (CU. FT.) QUANTITY CONTAINER SIZE (L x W x D) (INCHES) WEIGHT (LBS) DATE INTERIOR: **GROSS:** ITEM: EXTERIOR: As noted As noted 05 OCT 92 ITEM NAME MANUFACTURER Mail Tray Material As noted CONTAINER NAME CONTAINER COST PACK DESCRIPTION Plastic board (Type II - Laminated Corrugated) Ambient testing unless otherwise specified REF STD/SPEC TEST CONTAINER INSTRU-AND TEST METHOD OR TEST TITLE AND PARAMETERS NO. MENTATION ORIENTATION PROCEDURE NO'S 6. FLATNESS TEST MIL-P-83668A Specimens with dimensions Visual and 11 in. x 11 in. shall be Paragraph 4.5.8 dimensional cut from the middle of inspection the sheets. Condition at $+70^{\circ}F + 5^{\circ}F$ and 50% RH \pm 5% RH for a minimum of 24 hours. Place specimens on a flat surface and measure the maximum vertical distance from the surface to the underside of the specimen. 7. HUMIDITY TEST Visual and MIL-STD-810E Specimens shall be Method 507.3 11 in. x 11 in. dimensional Test at $+140^{\circ}F \pm 5^{\circ}F$ and inspection 90% RH \pm 5% RH for a minimum of 168 hours. COMMENTS: TEST NOS. 6 and 7, FLATNESS TEST and HUMIDITY TEST: Note that specimen size has been changed due to physical limitations. APPROVED BY: Larry A. Wood

Susan J. Misra, Materials Engineer

Materials Engineering, AFPEA

APPENDIX B

PHOTOGRAPHS



Fire the Standard-Type Mail Tray.

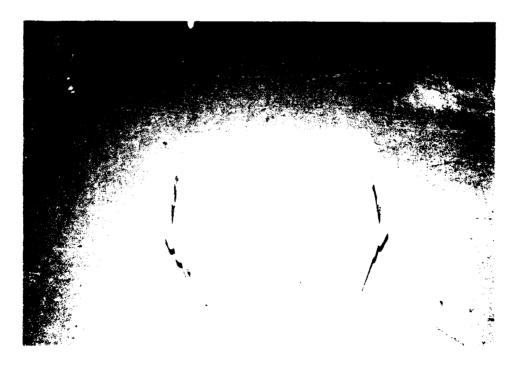


Figure 1.1 Chamier i Sype Mori Iray with Iri.

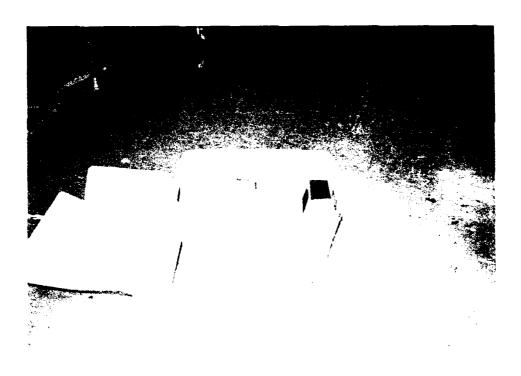


Figure 4. Test Lead of Bollb. of White Pond Equal.

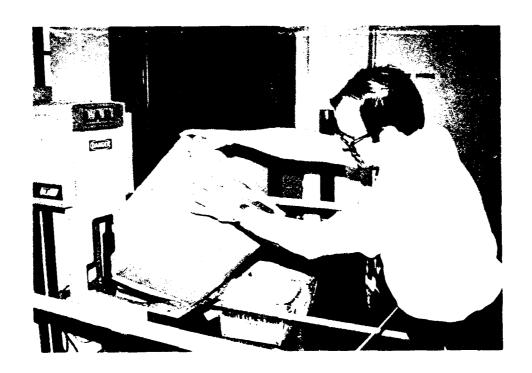


Figure 4. The shear with (3.4)



France 5. Edgewine Drop.



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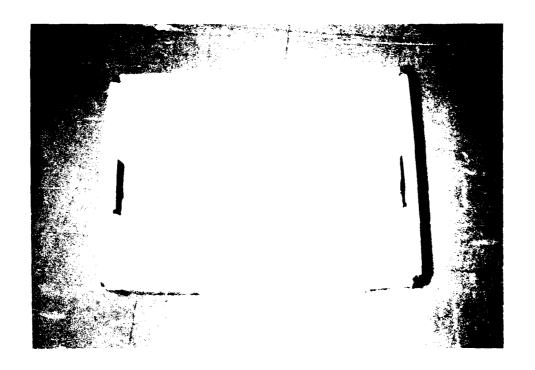


Figure 7. (C. A. Container Laiter In p. at +.46 E.

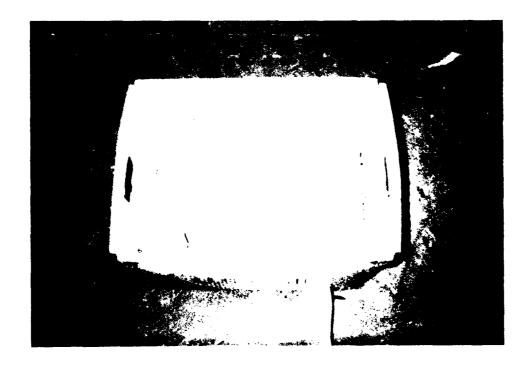
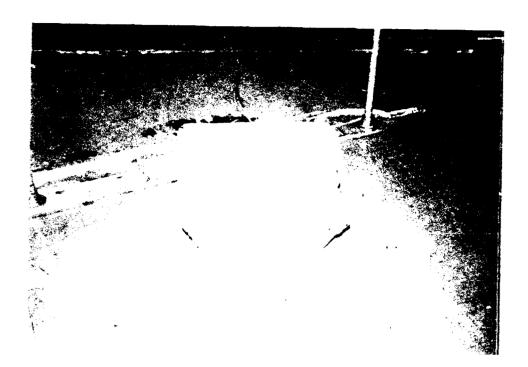


Figure 7. (a) A, Cabbashed Latter In p. at +.46 ft.



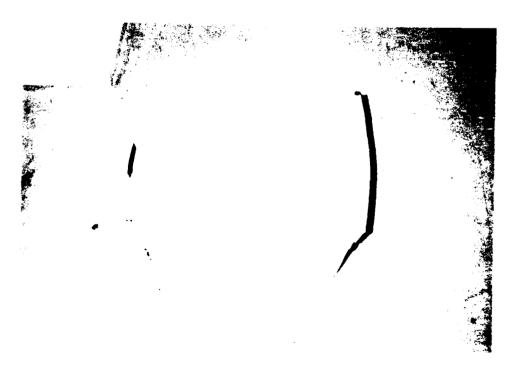




Figure 11. Co A, Container 2 after Drop on Corner To-T $^{\circ}$ +145 $^{\circ}F$.



Figure 12. Co B Container after Drop on Edge 1 8 at +145°F.



Figure 18. The first eigenvalues from the point as $\kappa = 4 \, \mathrm{GeV}$, $4 \, \mathrm{GeV}$

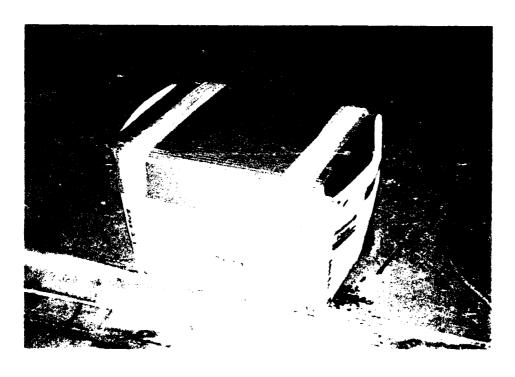
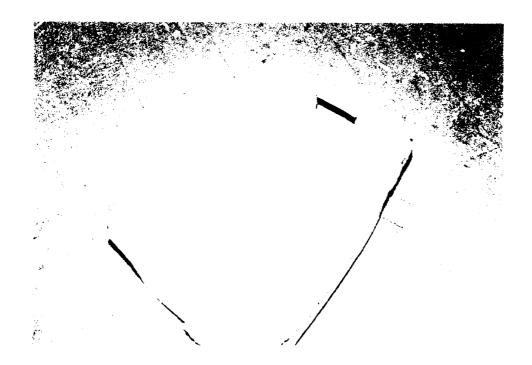


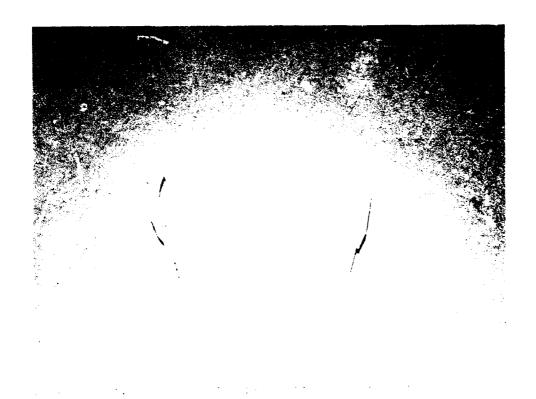
Figure 14. The Container wither trap on Corner 18.1. The ± 148 Fig.











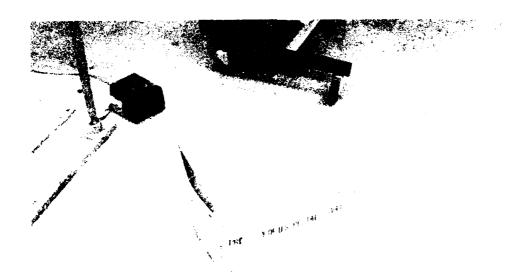
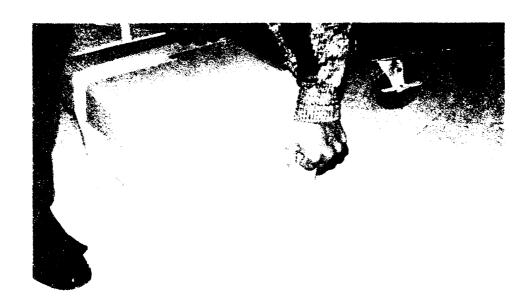




Figure 21. Co A, Container 1 after All Drops.

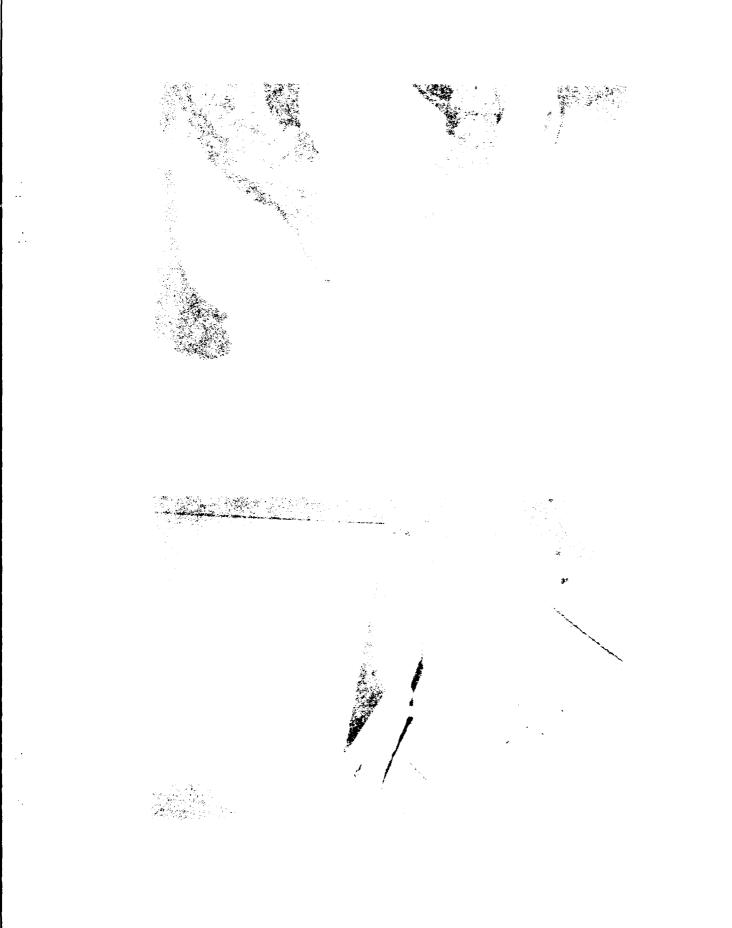


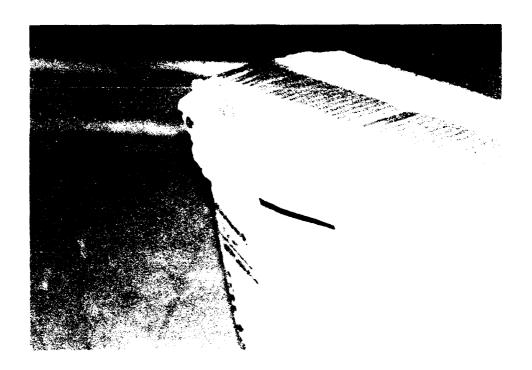
Figure 32. C. A. Container Claifer Drop on Edge 78 74 \pm 280F.





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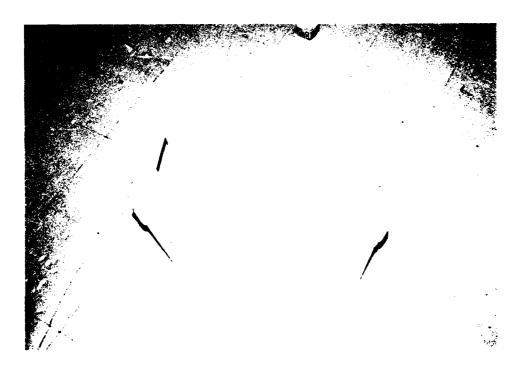
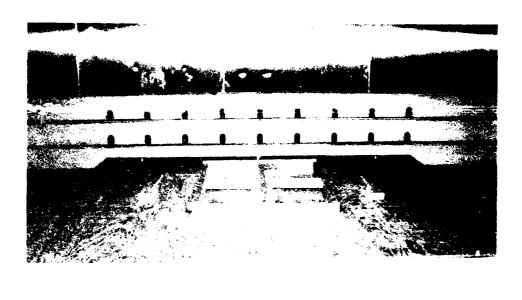


Figure 18. Tibration Test - Test Leak for Still Casi-





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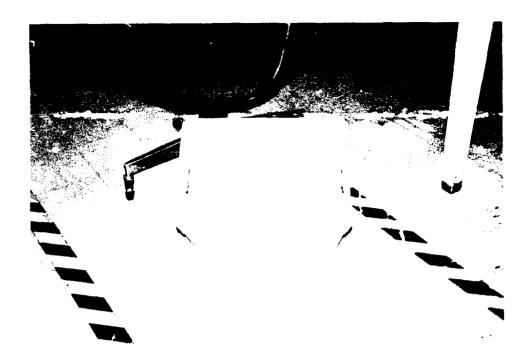
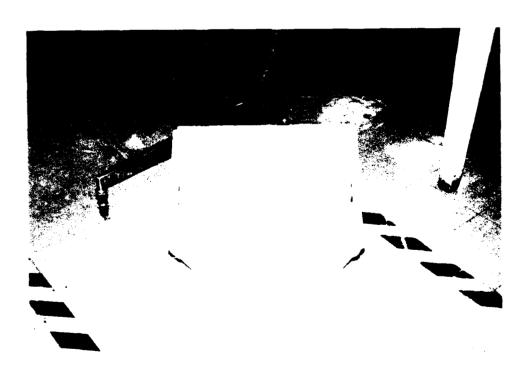


Figure 31. Co A, Container Lafter Tibration Test.



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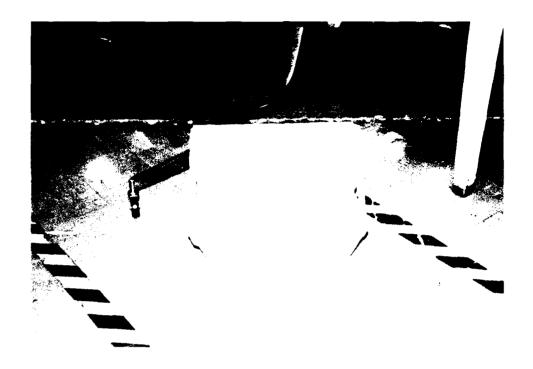


Figure 33. Co B Container after Vibration Test.

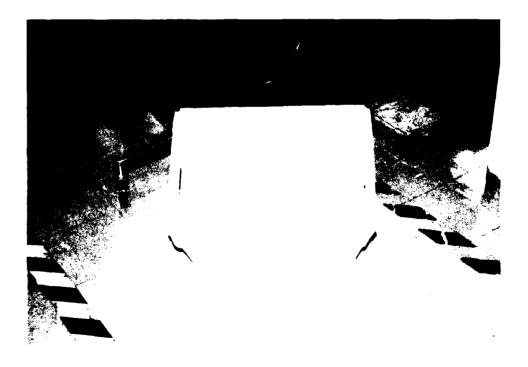
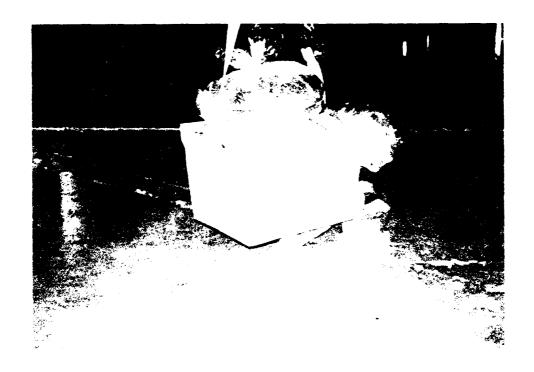
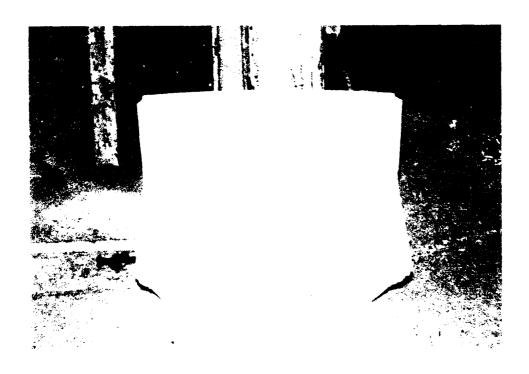


Figure 34. Co.C. Container after Militation Text.



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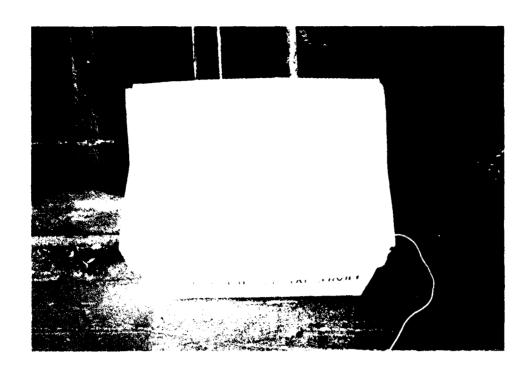


Figure 39. Co C Container with H 127 E. $t_{\rm c}$. That .

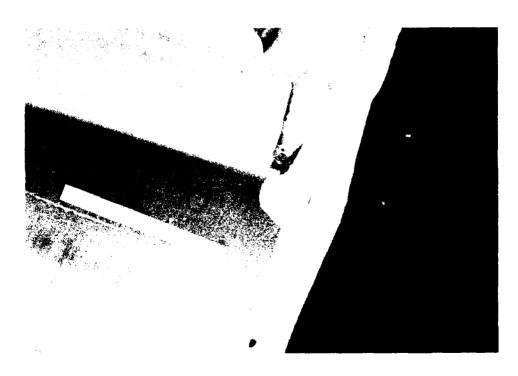
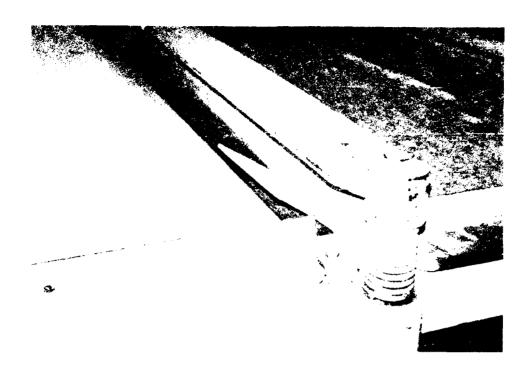
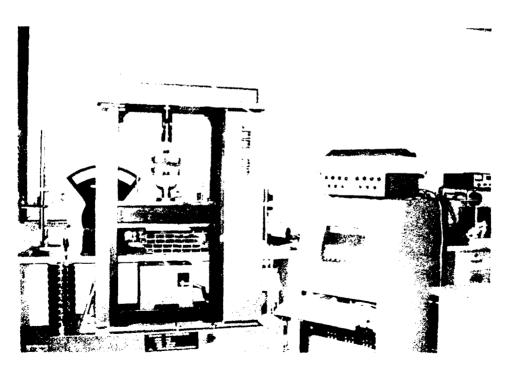


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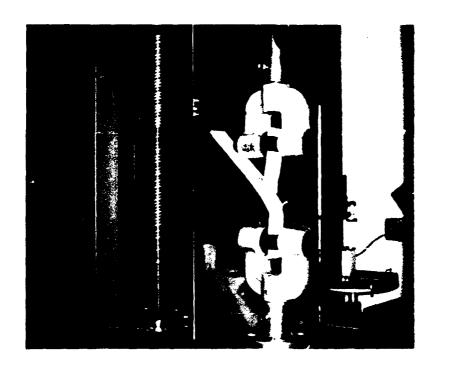


Figure 43. Flute T-Peel Strength Test.

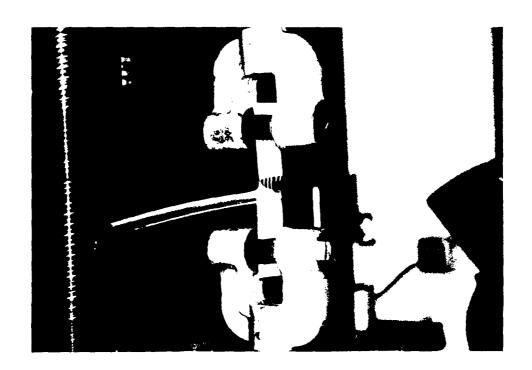


Figure 44. Co B Material Rupturing During the Flute T-Peel Strength Test.

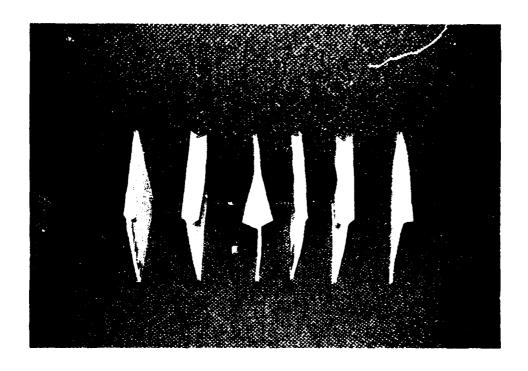


Figure 45. Co A, Material 1, Delamination which Octures after the Humidity Test.

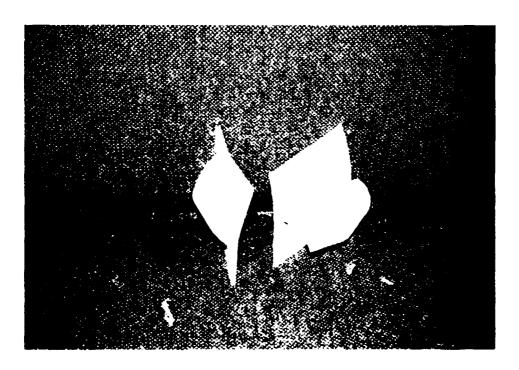
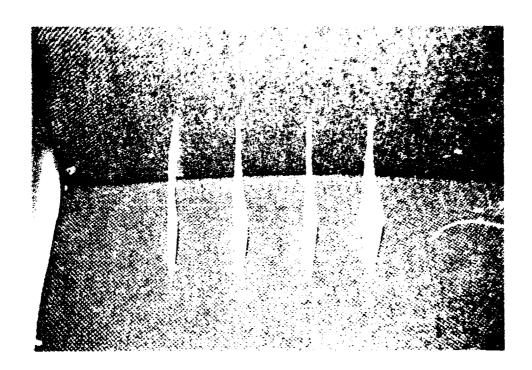
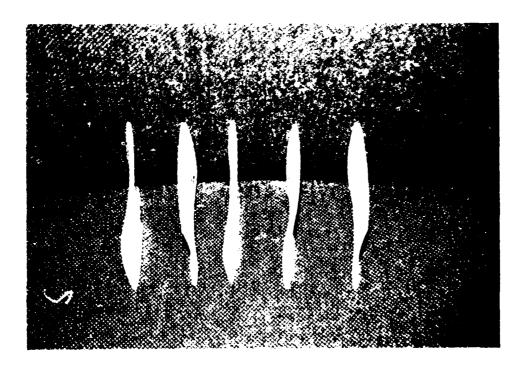


Figure 46. C. A. Material I. Sample: Delaminate Lty-Manually Fulling the Layer. Apart.



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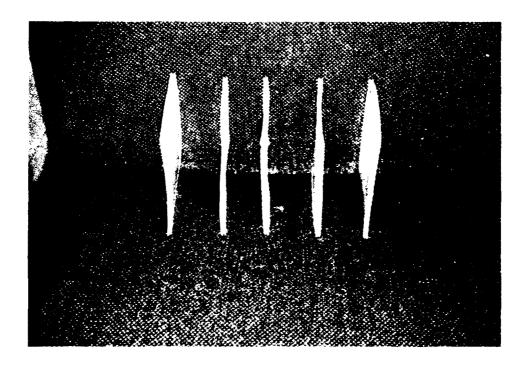


Figure 49. Company C Material after the Humidity Test.

APPENDIX C
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DISTRIBUTION LIST

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ASC/SDM WRIGHT-PATTERSON AFB OH 45433	1
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HQ ATC/LGTT RANDOLPH AFB TX 78150-5001	1
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APPENDIX D
REPORT DOCUMENTATION

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Unclassified 2a. SECURITY CLASSIFICATION AUTHORITY			3. DISTRIBUTION/AVAILABILITY OF REPORT								
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2b. DECLASSIFICATION / DOWNGRADING SCHEDULE				E	Approved for public release distribution unlimited						
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	ion Acti			HQ AFMC/LGTP							
6c. ADDRESS (City, State, and ZIP Code) HQ AFMC/LGTP 5215 Thurlow St					7b. ADDRESS (City, State, and ZIP Code)						
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					ELEMENT NO.	NO.			ACCESSION NO.		
11. TITLE (Incl	ude Security C	lassification)				<u> </u>	-				
	•	Mail Tray	y s								
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17.	COSATI	CODES	1	18. SUBJECT TERMS (C	(Continue on reverse if necessary and identify by block number)						
FIELD	GROUP	SUB-GROU	P	Mail Trays,	Mail Tray Material, Plastic Corrugated						
				Container, P	lastic Corr	ugated Mater	ial,	Mail, Co	ontainer		
19. ABSTRACT	(Continue on	reverse if nece	essary a	and identify by block nu	ımber)						
The Air Force Packaging Evaluation Activity (AFPEA), in response to a request from the United States Postal Service Engineering and Development Center (USPS EDC), conducted container testing and materials testing on mail trays and their plastic corrugated material, from three manufacturers. The trays were tested against FFD-SID-101, Methods 5007.1 and 5019.1 and FED-SID-648 A, Paragraph 5.8.3. The material was tested against MIL-P-83668A and MIL-SID-810E, Method 507.3. The mail trays were provided by the USPS EDC. The mail tray material was provided by the manufacturers.											
The test series was performed at the Air Force Packaging Evaluation Activity, 5215 Thurlow St, Wright—Patterson AFB, Ohio, 45433—5540.											
SUMMARY											
An evaluation was done on the performance of the mail trays and their material.											
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